

ductile materials (steel, stainless steel, aluminum, titanium, bronze)

distortion energy failure theory using maximum octahedral shear

von Mises equivalent stress σ'

$$\sigma' = \sqrt{\sigma_X^2 + \sigma_Y^2 - \sigma_X \sigma_Y + 3 \tau_{XY}^2} \quad \text{Eq 5.7d Norton}$$

normal stress and shear stress $\sigma_Y = 0$

$$\sigma' = \sqrt{\sigma_{\text{NORMAL}}^2 + 3 \tau_{\text{SHEAR}}^2}$$

normal stress and shear stress (round shaft with bending moment M and torsional moment T)

$$\sigma' = \sqrt{\sigma_{\text{BENDING}}^2 + 3 \tau_{\text{TORSION}}^2} \quad \sigma = \frac{M c}{I} \quad \tau = \frac{T c}{J} \quad c = \frac{d}{2} \quad I = \frac{\pi d^4}{64} \quad J = \frac{\pi d^4}{32}$$

$$\sigma' = \sqrt{\left(\frac{M c}{I}\right)^2 + 3 \left(\frac{T c}{J}\right)^2} = \sqrt{\left(\frac{64 M (d/2)}{\pi d^4}\right)^2 + 3 \left(\frac{32 T (d/2)}{\pi d^4}\right)^2} = \frac{32}{\pi d^3} \sqrt{M^2 + \frac{3}{4} T^2}$$

$$\sigma' = \frac{32}{\pi d^3} \sqrt{M^2 + \frac{3}{4} T^2}$$

biaxial normal stress (thin wall pressure vessel) $\tau_{XY} = 0$

$$\sigma' = \sqrt{\sigma_{\text{TANGENTIAL}}^2 + \sigma_{\text{AXIAL}}^2 - \sigma_{\text{TANGENTIAL}} \sigma_{\text{AXIAL}}}$$

pure shear (direct shear, key shear, shaft with no bending) $\sigma_X = \sigma_Y = 0$

$$\sigma' = \sqrt{3} \tau_{\text{SHEAR}}$$

brittle materials (cast iron, ceramics)

maximum tensile failure theory

Dowling equivalent stress $\tilde{\sigma}$

1) find principal normal stresses using Mohr's circle σ_1 σ_2 σ_3

2) calculate Dowling constants C_1 C_2 C_3 using Eq. 5.12c Norton $S_{UT} > 0$ $S_{UC} < 0$

$$C_1 = \frac{1}{2} \left[|\sigma_1 - \sigma_2| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_1 + \sigma_2) \right]$$

$$C_2 = \frac{1}{2} \left[|\sigma_2 - \sigma_3| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_2 + \sigma_3) \right]$$

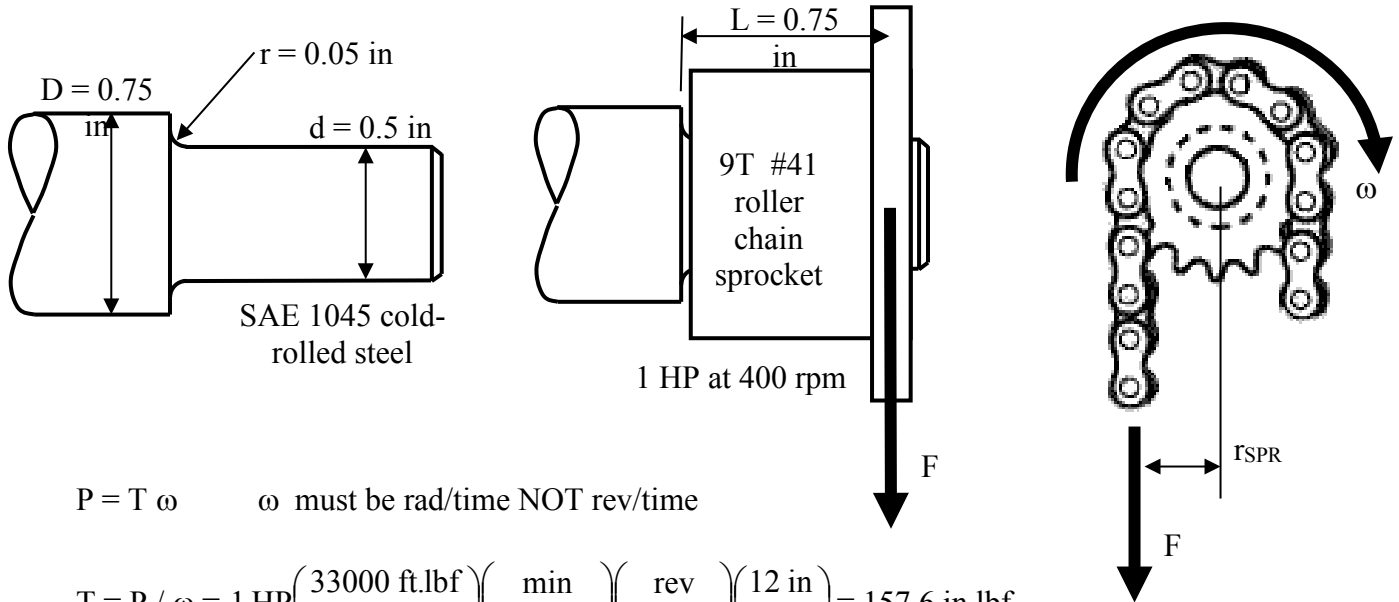
$$C_3 = \frac{1}{2} \left[|\sigma_3 - \sigma_1| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_3 + \sigma_1) \right]$$

3) find maximum

$$\tilde{\sigma} = \text{MAX}(C_1 C_2 C_3 \sigma_1 \sigma_2 \sigma_3) \quad \tilde{\sigma} = 0 \quad \text{if} \quad \text{MAX} < 0$$

4) compare Dowling equivalent stress to ultimate tensile strength

$$N = S_{UT} / \tilde{\sigma}$$



$$P = T \omega \quad \omega \text{ must be rad/time NOT rev/time}$$

$$T = P / \omega = 1 \text{ HP} \left(\frac{33000 \text{ ft.lbf}}{\text{HP.min}} \right) \left(\frac{\text{min}}{400 \text{ rev}} \right) \left(\frac{\text{rev}}{2\pi \text{ rad}} \right) \left(\frac{12 \text{ in}}{\text{ft}} \right) = 157.6 \text{ in.lbf}$$

#41 ANSI chain, pitch $p = 0.5$ inch, maximum working load = 314 lbf, McMaster-Carr

$$\text{pitch circumference} = 2 \pi r_{\text{SPR}} \approx (9T) p$$

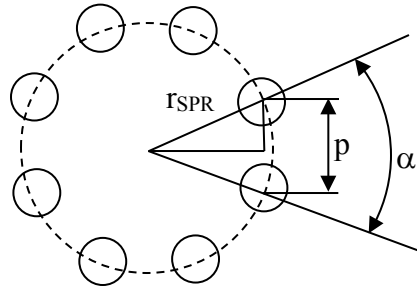
$$r_{\text{SPR}} \approx (9T) p / 2 \pi = 0.7162 \text{ in}$$

$$\alpha = 360^\circ / \text{number of teeth} = 40^\circ$$

$$p = 0.5 \text{ in}$$

$$\sin \alpha/2 = (p/2) / r_{\text{SPR}}$$

$$\text{EXACT } r_{\text{SPR}} = (p/2) / (\sin \alpha/2) = 0.731 \text{ in}$$



$$T = F r_{\text{SPR}} \quad F = T / r_{\text{SPR}} = 220 \text{ lbf}$$

$$\text{bending moment } M = F L = 165 \text{ in.lbf}$$

$$\text{torsion moment } T = 157.6 \text{ in.lbf}$$

$$I = \pi d^4 / 64 = 3.068 \times 10^{-3} \text{ in}^4 \quad J = \pi d^4 / 32 = 6.136 \times 10^{-3} \text{ in}^4$$

$$c = d / 2 = 0.25 \text{ in}$$

$$\sigma_{\text{NOM}} = M c / I = \frac{(165 \text{ in.lbf})(0.25 \text{ in})}{(3.068 \times 10^{-3} \text{ in}^4)} = 13.45 \text{ ksi}$$

$$\tau_{\text{NOM}} = T c / J = \frac{(157.6 \text{ in.lbf})(0.25 \text{ in})}{(6.136 \times 10^{-3} \text{ in}^4)} = 6.421 \text{ ksi}$$

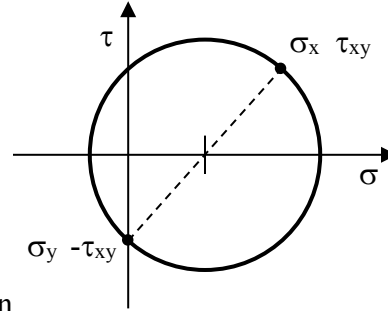
maximum shear failure criterion (OLD)

$$\sigma_x = 13.45 \text{ ksi} \quad \sigma_y = 0 \quad \tau_{xy} = 6.421 \text{ ksi}$$

$$\tau_{MAX} = \sqrt{(\sigma_x / 2)^2 + \tau_{xy}^2} = 9.30 \text{ ksi}$$

$$1045 \text{ cold-rolled steel} \quad S_y = 77 \text{ ksi} \quad \text{Table A-9 Norton}$$

$$N = \frac{0.5 S_y}{\tau_{MAX}} = 4.14 \quad \text{Eq 5.10 and 5.11 Norton}$$

**von Mises failure criterion**

$$\sigma_x = 13.45 \text{ ksi} \quad \sigma_y = 0 \quad \tau_{xy} = 6.421 \text{ ksi}$$

$$\sigma' = \sqrt{\sigma_x^2 + \sigma_y^2 - \sigma_x \sigma_y + 3\tau_{xy}^2} = 17.45 \text{ ksi} \quad \text{Eq 5.7d Norton}$$

$$1045 \text{ cold-rolled steel} \quad S_y = 77 \text{ ksi} \quad \text{Table A-9 Norton}$$

$$N = \frac{S_y}{\sigma'} = 4.41 \quad \text{Eq 5.8a Norton}$$

iterative solution for new diameter

$$\sigma_{NOM} = M c / I = \frac{M \frac{d}{2} 64}{\pi d^4} = \frac{32 M}{\pi d^3} \quad \tau_{NOM} = T c / J = \frac{T \frac{d}{2} 32}{\pi d^4} = \frac{16 T}{\pi d^3}$$

try NEW $d = 0.625$ in

$$\text{NEW } \sigma' = \text{OLD } \sigma' \left(\frac{0.5}{0.625} \right)^3 = 0.512 \text{ OLD } \sigma'$$

static stress concentration factors

$$\text{shaft bending} \quad D/d = 1.5 \quad A = 0.93836 \quad b = -0.25759 \quad \text{Figure C-2 Norton}$$

$$r/d = 0.1 \quad K_t = A(r/d)^b = 1.698$$

$$\sigma = K_t \sigma_{\text{NOM}} = 22.84 \text{ ksi}$$

shaft torsion $D/d = 1.5$ interpolated $A = 0.8526$ $b = -0.2334$ Figure C-3 Norton

$$r/d = 0.1 \quad K_t = A(r/d)^b = 1.459$$

$$\tau = K_t \tau_{\text{NOM}} = 9.368 \text{ ksi}$$

on-line calculator

<https://www.amesweb.info/StressConcentrationFactor/SteppedShaftWithShoulderFillet.aspx>

bending $K_t = 1.74$ torsion $K_t = 1.41$

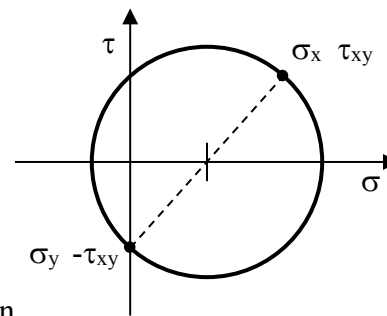
maximum shear failure criterion (OLD)

$$\sigma_x = 22.84 \text{ ksi} \quad \sigma_y = 0 \quad \tau_{xy} = 9.368 \text{ ksi}$$

$$\tau_{\text{MAX}} = \sqrt{(\sigma_x / 2)^2 + \tau_{xy}^2} = 14.77 \text{ ksi}$$

1045 cold-rolled steel $S_y = 77 \text{ ksi}$ Table A-9 Norton

$$N = \frac{0.5 S_y}{\tau_{\text{MAX}}} = 2.61 \quad \text{Eq 5.10 and 5.11 Norton}$$



von Mises failure criterion

$$\sigma_x = 22.84 \text{ ksi} \quad \sigma_y = 0 \quad \tau_{xy} = 9.368 \text{ ksi}$$

$$\sigma' = \sqrt{\sigma_x^2 + \sigma_y^2 - \sigma_x \sigma_y + 3\tau_{xy}^2} = 28.02 \text{ ksi} \quad \text{Eq 5.7d Norton}$$

1045 cold-rolled steel $S_y = 77 \text{ ksi}$ Table A-9 Norton

$$N = \frac{S_y}{\sigma'} = 2.75 \quad \text{Eq 5.8a Norton}$$

key length

assume 1020 hot-rolled $S_y = 30 \text{ ksi}$ Table A-9 Norton

$d = 0.5$ inch DIA shaft $w = 0.125$ inch square key Table 10-2 Norton

assume $L_{KEY} = 0.625$ inch $< L$ overhang

$$T = 157.6 \text{ in.lbf} \quad r_{SHAFT} = 0.25 \text{ inch} \quad F_{KEY} = T / r_{SHAFT} = 630.4 \text{ lbf}$$

$$A_{SHEAR} = w L = 0.078125 \text{ in}^2 \quad \tau = F_{KEY} / A_{SHEAR} = 8069.1 \text{ psi}$$

$$\sigma' = \sqrt{3} \tau = 13.98 \text{ ksi}$$

$$N_{YIELD} = S_Y / \sigma' = 2.15$$

fatigue stress concentration (notch sensitivity) factors

full reversed bending $\sigma_{NOM} = 13.45$ ksi $K_t = 1.698$

1045 cold-rolled steel $S_{ut} = 91$ ksi Table A-9 Norton

Neuber $\sqrt{a} = 0.0692 \sqrt{\text{in}}$ interpolated Table 6-6 Norton

$$q = \frac{1}{1 + \frac{\sqrt{a}}{\sqrt{r}}} = \frac{1}{1 + \frac{0.0692\sqrt{\text{in}}}{\sqrt{0.05 \text{ in}}}} = 0.7637 \quad \text{Eq 6.13 Norton}$$

$$K_f = 1 + q (K_t - 1) = 1.533 \quad \text{Eq 6.11b Norton}$$

$$\sigma = K_f \sigma_{NOM} = 20.62 \text{ ksi}$$

$$\sigma_{alt}' = \sigma = 20.62 \text{ ksi}$$

constant torsion $\tau_{NOM} = 6.421$ ksi $K_t = 1.459$

no fatigue stress concentration (notch sensitivity) factor $\tau = K_t \tau_{NOM} = 9.368$ ksi

$$\sigma_{mean}' = \sqrt{3} \tau = 16.23 \text{ ksi} \quad \text{Eq 5.7d Norton}$$

modified endurance limit

1045 cold-rolled steel $S_{ut} = 91$ ksi Table A-9 Norton

$$S_e' = 0.5 S_{ut} = 45.5 \text{ ksi} \quad \text{Eq 6.5a Norton}$$

$$C_{LOAD} = 1.0 \quad \text{bending} \quad \text{Eq 6.7a Norton}$$

$$C_{\text{SIZE}} = 0.869 d^{-0.097} = 0.9294 \quad \text{circular section} \quad \text{Eq 6.7b Norton}$$

$$C_{\text{SURF}} = 0.76 \quad \text{machined} \quad \text{Figure 6-26 Norton}$$

$$C_{\text{SURF}} = A(S_{\text{ut}})^b = 0.817 \quad A = 2.7 \quad b = -0.265 \quad \text{machined} \quad \text{Eq 6.7e and Table 6-3 Norton}$$

$$\text{use } C_{\text{SURF}} = 0.76$$

$$C_{\text{TEMP}} = 1.0 \quad \text{normal environmental temperature range} \quad \text{Eq 6.7f Norton}$$

$$C_{\text{RELI}} = 0.814 \quad R = 99\% \quad \text{Table 6-4 Norton}$$

$$S_e = C_{\text{LOAD}} C_{\text{SIZE}} C_{\text{SURF}} C_{\text{TEMP}} C_{\text{RELI}} S_e' = (1) (0.9294) (0.76) (1) (0.814) 45.5 \text{ ksi} = 26.16 \text{ ksi}$$

fatigue factor of safety

$$\left(\frac{400 \text{ rev}}{\text{min}} \right) \left(\frac{60 \text{ min}}{\text{hr}} \right) \left(\frac{24 \text{ hr}}{\text{day}} \right) = 576,000 \text{ rev/day}$$

$$\text{use infinite life} \quad N > 10^6 \quad S_N = S_e = 26.16 \text{ ksi} \quad S_{\text{ut}} = 91 \text{ ksi}$$

$$\sigma_{\text{alt}}' = 20.62 \text{ ksi} \quad \sigma_{\text{mean}}' = 16.23 \text{ ksi}$$

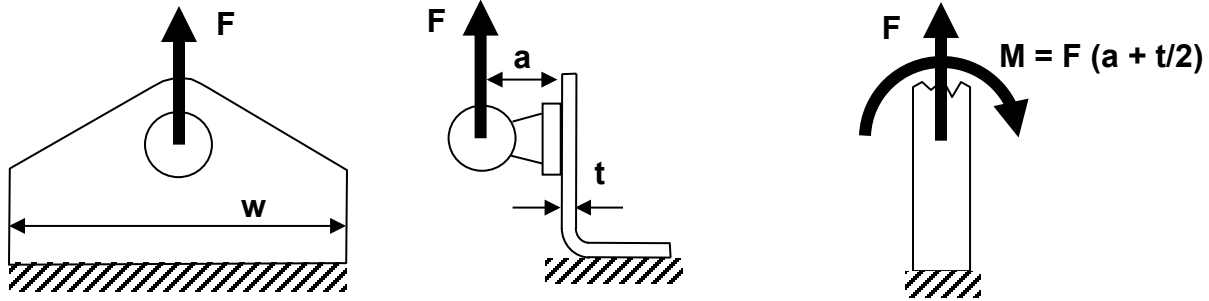
$$N_{\text{alt}} = S_N / \sigma_{\text{alt}}' = 1.269 \quad N_{\text{mean}} = S_{\text{ut}} / \sigma_{\text{mean}}' = 5.607$$

$$N_{\text{GOODMAN}} = N_{\text{alt}} N_{\text{mean}} / (N_{\text{alt}} + N_{\text{mean}}) = 1.03$$

$$1045 \text{ cold-rolled steel} \quad S_y = 77 \text{ ksi} \quad \text{Table A-9 Norton}$$

$$N_{\text{FIRST}} = S_y / (\sigma_{\text{alt}}' + \sigma_{\text{mean}}') = 2.09$$

A ball mount bracket for gas springs is made from AISI 1010 hot rolled steel as shown below. Determine vertical load F on the ball that would cause static yielding.



$$w = 50.8 \text{ mm} \quad t = 2 \text{ mm} \quad a = 12.7 \text{ mm}$$

AISI 1010 hot-rolled steel $S_y = 179 \text{ MPa}$ Table A-9 Norton

$$\sigma = \frac{Mc}{I} + \frac{F}{A} \quad \text{at front face}$$

$$c = t/2 = 1 \text{ mm} \quad I = w t^3 / 12 = 33.87 \text{ mm}^4 \quad A = w t = 101.6 \text{ mm}^2 \quad (a + t/2) = 13.7 \text{ mm}$$

use $\sigma = S_y$ for yield

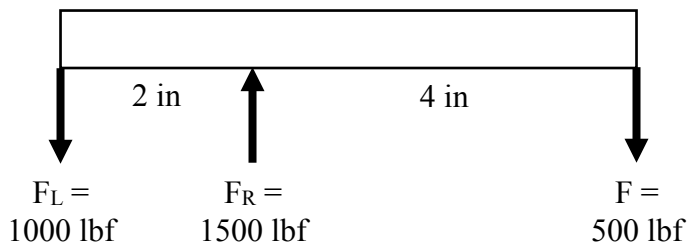
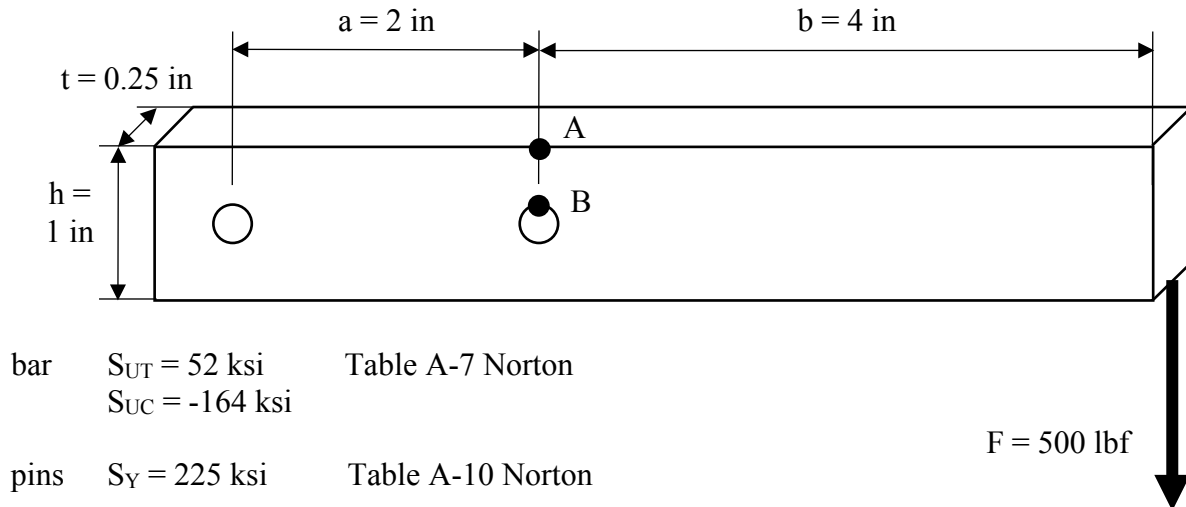
$$S_y = F \left(\frac{(a + t/2)c}{I} + \frac{1}{A} \right)$$

$$179 \frac{\text{N}}{\text{mm}^2} = F \left(\frac{(13.7 \text{ mm})(1 \text{ mm})}{33.87 \text{ mm}^4} + \frac{1}{101.6 \text{ mm}^2} \right) = F(0.4045 + 0.009843) \frac{1}{\text{mm}^2}$$

$$F = 432.0 \text{ N} = 97.12 \text{ lbf}$$

Class 50 cast iron bar mounted with two 0.25 inch DIA dowel pins (A8 steel hardness HRC52)

values shown in red are for 0.5 inch DIA dowel pins



direct shear on right pin

$$A = \pi d^2 / 4 = 0.0491 \text{ in}^2$$

$$0.1963 \text{ in}^2$$

$$\tau = F_R / A = 30.55 \text{ ksi}$$

$$7.639 \text{ ksi}$$

$$\sigma' = \sqrt{3} \tau = 52.91 \text{ ksi}$$

$$13.23 \text{ ksi}$$

$$N = S_Y / \sigma' = 4.25$$

$$17.00$$

bending of bar at top A

maximum M at right pin

$$M = (500 \text{ lbf}) (4 \text{ in}) = 2000 \text{ in.lbf}$$

$$I = t (h^3 - d^3) / 12 = 0.0205 \text{ in}^4$$

$$0.01823 \text{ in}^4$$

$$c = h / 2 = 0.5 \text{ in}$$

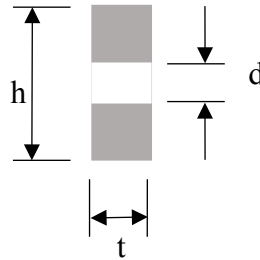
$$\sigma = M c / I = 48.78 \text{ ksi}$$

$$54.85 \text{ ksi}$$

$$\tilde{\sigma} = \sigma$$

$$N = S_{UT} / \tilde{\sigma} = 1.07$$

$$0.95$$



direct crushing of bar above right pin B

assume tight fit $A_{PROJ} = d t = 0.0625 \text{ in}^2$ Section 4.8 Norton

$$0.125 \text{ in}^2$$

$$\sigma_{CRUSH} = F_R / A_{PROJ} = -24.00 \text{ ksi} \quad \text{compressive}$$

$$-12.00 \text{ ksi}$$

bending of bar above right pin B

$c = d / 2 = 0.125 \text{ in}$ same M same I

$$0.25$$

$$\sigma_{BEND} = M c / I = +12.20 \text{ ksi} \quad \text{tension}$$

$$+27.43 \text{ ksi}$$

Dowling equivalent stress Eq. 5.12c Norton

$$\sigma_1 = +12.20 \text{ ksi} \quad \sigma_2 = 0 \quad \sigma_3 = -24.00 \text{ ksi}$$

$$27.43 \text{ ksi} \quad 0 \quad -12.00 \text{ ksi}$$

$$S_{UT} = 52 \text{ ksi} \quad S_{UC} = -164 \text{ ksi}$$

$$\frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} = 0.3659$$

$$C_1 = \frac{1}{2} \left[|\sigma_1 - \sigma_2| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_1 + \sigma_2) \right] = +8.33 \text{ ksi}$$

$$18.73 \text{ ksi}$$

$$C_2 = \frac{1}{2} \left[|\sigma_2 - \sigma_3| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_2 + \sigma_3) \right] = +7.61 \text{ ksi}$$

$$3.80 \text{ ksi}$$

$$C_3 = \frac{1}{2} \left[|\sigma_3 - \sigma_1| + \frac{2 S_{UT} - |S_{UC}|}{-|S_{UC}|} (\sigma_3 + \sigma_1) \right] = +15.94 \text{ ksi} \quad 22.54 \text{ ksi}$$

$$\tilde{\sigma} = \text{MAX}(C_1, C_2, C_3, \sigma_1, \sigma_2, \sigma_3) = +15.94 \text{ ksi} \quad \text{Note: not MAX(abs())} \quad 27.43 \text{ ksi}$$

$$N = S_{UT} / \tilde{\sigma} = 3.26 \quad 1.90$$